### Aerospace Composite Materials

### E-720 Epoxy Prepregs

Park's E-720 is a modified epoxy resin system designed to retain excellent mechanical properties after extended exposure to high temperature. E-720 has a proven history in many demanding aerospace applications.

### **Key Features & Benefits**

- Excellent retention of mechanical properties after longterm high temperature exposure
- Long out-time for easy processing
- Good electrical properties

#### **Product Forms**

- Available on a wide variety of reinforcements including fiberglass and quartz
- Solution coated fabrics up to 60 inches wide
- Compatible with Autoclave or Press Molding processes

### **Product Overview**

### **Applications / Qualifications**

- Secondary Aircraft Structures
- Radomes
- Nacelles
- Inlet Ducts
- Fairings

#### For Information about Park's materials:

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## Aerospace Composite Materials

### Technical Datasheet

### E-720 Epoxy Prepregs

### **Prepreg and Laminate Physical Properties**

Reinforcement	7781 E-Glass	581 Quartz
Fabric Area Weight (gsm)	300	292
Prepreg Resin Content (%)	33 – 37	33 – 37
Resin Flow (325°F, 50 psi) (%)	10 – 24	10 – 24
Volatiles (275°F) (% max)	2.0	2.0
Gel Time (min)	1 – 2	1 – 2
Laminate Tg – std cure (°C)	180	180
Laminate Tg – post cure (°C)	230	230
Barcol Hardness	80	
Dielectric Constant (Dk)	4.2	3.3 – 3.6
Loss Tangent (Df)	0.015	0.012 - 0.014

### **Processing Guidelines**

### **Prepreg Storage Life**

Tack Life: 14 days @ 75°F Out Life: 30 days @ 75°F Shelf Life: 6 months @ 0°F <u>Note</u>: The following guidelines are provided to assist Park material users with general recommendations for successful processing. The recommendations are for general review purposes only and process adjustments may be required to achieve optimum results in your specific manufacturing environment.

### **Autoclave Cure Cycle**

- Apply 24"Hg vacuum (minimum) for 1 hour before beginning heat cycle
- Apply 60psi autoclave pressure.
- Raise product temperature from RT to 230°F at 2 3°F/min
- After 30 minute hold time at 230°F, continue ramp to 350°F
- Hold product at cure temperature for 2 hours
- Cool product to 150°F at no more than 8°F/min
- Recommended post-cure: 1 hours at 500°F or 4 hours at 400°F

All test data provided are typical values and not intended to be specification values. For review of critical specification tolerances, please contact a company representative directly.

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# Aerospace Composite Materials

### Technical Datasheet

# **E-720 Epoxy Prepregs**

### **Laminate Mechanical Properties**

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Reinforcement		7781 E-glass	581 Quartz
Tensile Strength, 0° (Ksi)			
75°F	Dry	60	90
350°F	Dry	50	
420°F	Dry	45	60
500°F	Dry	45	
ASTM-D-638			
Compressive	Compressive Strength (Ksi)		
75°F	Dry	60	50
350°F	Dry	52	
420°F	Dry	52	30
500°F	Dry	45	
ASTM-D-695			
Flexural Stren			
75°F	Dry	85	90
350°F	Dry	60	
420°F	Dry	50	40
500°F	Dry	30	
ASTM-D-790			
Flexural Modu			
75°F	Dry	3.4	3.4
350°F	Dry	2.9	
420°F	Dry	2.8	2.0
500°F	Dry	2.4	
ASTM-C-790	hear Ctonath (Va:)		
	hear Stength (Ksi)	6.7	
75°F	Dry	6.7	
350°F ASTM-D-2344	Dry	4.8	
A311VI-D-2344			

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